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Capturing Upstream Turbulence Information by Modeling the Turbulent Generation Region and Comparison of 0° and 30° Flow and Dispersion CFD Modeling and Macdonald's MUST Experiment

John P. Keady¹, Pablo Huq², W.G.K. Bradbury¹, P.E. Price¹ and Fernando E. Camelli¹

¹SPACS, Center for Computational Fluid Dynamics, George Mason University ²College of Earth, Ocean & Environment, University of Delaware





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Objectives:

•Simulate the Laboratory MUST Experiment

•More accurately model pre-test section turbulent structure of velocity flow by modeling the roughness section of the experimental tunnel

•Comparison of Simulation and Experimental parameters

Parameters Examined: Velocity in x and z-direction Turbulent Kinetic Energy Normalized Temperature profile of injected plume



(2002/2003) (0 and 30 degree angle)

The MUST test array consisted of 50-mm high obstacles with a 4:1 (width-to-height) aspect ratio and had a packing density of 13.3% in a 12.8-m long hydraulic flume at the University of Waterloo, which has a 1.2-m x 1.2-m x 2.4-m long test section enclosed in tempered glass.





(2002/2003) (0 and 30 degree angle)



The experiment measured the flow characteristics of three incident angles 0, 30 and 45 degrees, where experimentally the velocity vector was kept constant and the array rotated.

Simulations were performed for two angles 0 and 30 degrees.



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(Flume Injection – Temperature Measurement)





Flume injections were compared for two experimental cases of flue injection:

In Array Before Gap – IABG In Array Before Object - IABO

<u>Temperature Measurements</u> The Plume Injection Temperature was 65°C while the ambient fluid before injection was 20°C. Temperature was measured using Resistance Temperature Detectors (RTD) (a thin film platinum detector type manufactured by OMEGA)





CFD Geometry Representation – IABO



• Mesh: 39 Million tetrahedra

Surface Mesh

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- Counihan spikes
- Short barrier wall
- Billboard
- Lego[®]

Cut Plane of Volume Mesh





Behind Object: Comparison of Experimental and Simulation Upstream Profiles



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u -velocity and Turbulent Kinetic Energy Comparison IABO – 0 Degrees







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u and v-velocity and Turbulent Kinetic Energy 0 and 30 degrees IABO – In Array Before Object Row 6 Value Comparisons







SPACS School of Physics, Astronomy, and Computational Sciences |V| m/s

0.13

0.10

0.06 0.03

0.00

IABO - 30 degrees







IABO - 30 degrees

SPACS School of Physics, Astronomy, and Computational Sciences |V| m/s

0.13

0.10

0.06 0.03

0.00



Summary Plots Row 6, Turbulent Kinetic Energy





Summary Plots

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Dispersion Comparison IABG



Comparison of lateral normalized temperature profiles of a plume IAB<u>G</u>, 0 degrees, rows 1, 2, and 3





Comparison of vertical normalized temperature profiles of a plume IABG, 0 degrees, rows 1, 2, and 3





Comparison of lateral normalized temperature profiles of a plume <u>IABG</u>, 0 degrees, row 3, for heights z/H = 0.1, 0.6, 1.0, and 2.4





Summary Plots

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Flow Comparison IABG and IABO 0 and 30 Degrees



Summary Plot: IABG vs IABO lateral Profile

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IABG Experiment



IABG Simulation





IABO Simulation







IABO – 0 deg – Temperature Iso-surface: T= 20.01 °C





IABG – 0 deg – Temperature Iso-surface: T= 20.01 °C





IABO – 30 deg – Temperature Iso-surface: T= 20.01 °C





Conclusions

• Modeling flow development through the roughness section provides a pre-test section velocity profile that maintains turbulence information, eliminating the need for independent turbulence modeling of the pre-test section velocity.

(Support: Modeled Uave vs. Exp Uave closely match) (Support: Modeled TKE vs. Exp TKE closely match)

 Modeling flow development through the roughness section provides a pre-test section velocity profile that maintains turbulence information, reproducing dispersion trends seen in experiments.

(Support: Modeled TN vs. Exp TN (closely follow similar trends such as peaks, extent))

• NEEDS: From Experimentalist:

For future simulation comparisons, experimental information is needed in the pre roughness section, and more details of pre-test section measurement positions are needed for simulation.

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EXTRA SLIDES



Upstream Flow Sensitivity

Summary Plots





Upstream Flow Sensitivity

Summary Plots





Time snapshots

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0 deg - IABG Simulation

^{V m/s} 0.13	
0.10	
0.06	
0.03	- The
0.00	Sec.

0 deg - IABO Simulation



30deg - IABO Simulation





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Comparison of normalized u-velocity behind rows 1, 3, and 6

Behind Object: Comparison of Mean Velocity Profiles Behind Rows 1, 3, and 6; 4-Profile Average





Mock Urban Setting MUST Experiment

Review

The MUST experiment was designed to represent an urban layout with symmetric characteristics. An array of 10 by 12 containers was placed at the U.S. Army Dugway Proving Ground Horizontal Grid test site in Utah. Each container was 12.2 m long, 2.42 m wide and 2.54 m high. The test area encompassed 280 m in width, and 50 m in height above the ground...

." 43rd AIAA Aerospace Sciences Meeting and Exhibit, January 10-13, Reno, Nevada, Fernando E. Camelli et al. .."







(2002/2003) (0 and 30 degree angle)

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Scale models were placed on the floor of the test section upstream of which a 800-mm deep, fully turbulent simulated atmospheric boundary layer (ABL) flow is created using a combination of a short barrier wall, turbulence spires, and an extended fetch of surface roughness. The surface roughness has two fetches. The first consists of 2.4-m of 19-mm high aluminum plates with packing density λ f=0.063. This is followed by a2.4-mfetch of 10-mm high x l6-mm wide Lego@ blocks on 48-mm centers with a packing density λ f=0.069.



(Velocity Measurement)

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Velocity Measurements

Velocity measurements in the flume are made with a three-dimensional SonTek micro-acoustic Doppler velocimeter (ADV) sampling at20 Hz. The ADV probe focuses a I6-MHz acoustic beam into a small control volume located 50-mm below the probe tip and uses the Doppler reflection from tiny particles in the flow to calculate the velocity.





Navier-Stokes Equations - CFD

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• Conservation of Mass

$$\nabla \cdot \vec{u} = 0$$

Conservation of Momentum

$$\rho \left[\frac{\partial \vec{u}}{\partial t} + \vec{u} \cdot \nabla \vec{u} \right] = \rho f_e - \nabla p + \mu \nabla^2 \vec{u}$$

• Temperature

$$\rho c_p \left[\frac{\partial T}{\partial t} + \vec{u} \cdot \nabla T \right] = \kappa \nabla^2 T$$

- Large Eddy Simulation
- Smagorinsky WALE Turbulence model



CFD Geometry Representation – IABG



SPACS School of Physics, Astronomy, and **Computational Sciences** CFD Geometry Representation – 30 deg IABO



Summary Plots

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Velocity and Structure Upstream of Test Section



Behind Object: Comparison of Experimental and Simulation Upstream Profiles





Comparison of Turbulent u-Intensities at Entrance of Test Section





Comparison of Turbulent v-Intensities at Entrance of Test Section Computational Sciences



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Comparison of Turbulent w-Intensities at Entrance of Test Section





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Row 6 Velocity Symmetry and Averaging Comparisons



Comparison of normalized u-velocity behind row 6









Comparison of normalized u-velocity behind row 6, line and area average





Comparison of normalized v-velocity behind row 6, line and area average

Behind Object: v-velocity values, row 6, line and area experiment and simulation averages





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IABO Rows 1, 3, and 6 Velocity and Turbulence



Comparison of normalized u-turbulent intensity behind rows 1, 3, and 6





Comparison of normalized v-turbulent intensity behind rows 1, 3, and 6

Behind Object: Comparison of Turbulence Intensity; Rows 1, 3, and 6; 4-Profile Ave





Comparison of normalized w-turbulent intensity behind rows 1, 3, and 6

Behind Object: Comparison of Turbulence Intensity; Rows 1, 3, and 6; 4-Profile Ave





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IABO Rows 1, 3, and 6 Turbulent Kinetic Energy



Behind Object: Comparison of Turbulence Kinetic Energy; Row 1





Behind Object: Comparison of Turbulence Kinetic Energy; Row 3





Comparison of normalized turbulence kinetic energy profiles behind 6

Behind Object: Comparison of Turbulence Kinetic Energy; Row 6





Summary Plots

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Dispersion Comparison IABO



Comparison of lateral normalized temperature profiles of a plume <u>IABO</u>, 0 degrees, row 3, for heights z/H = 0.1, 0.6, 1.0, and 2.4

