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11.2 ABI's Unique Calibration and Validation Capabilities

Dr. Paul C. Griffith

12th Annual Symposium on New Generation Operational Environmental Satellite Systems

AMS Annual Meeting, 11-14 January 2016

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Accurate radiometric calibration essential for reliable weather data products

Advanced imagers provide improved imagery but pose calibration challenges

Harris' ABI-class imagers provide calibration solutions

- Better on-board targets
- Unprecedented operational flexibility permitting improved on-orbit radiometric calibration utilizing external sources
 - e.g. Earth, Moon, ...
- Significantly more calibration opportunities
 - Ability to perform most calibrations <u>without</u> interrupting routine Full Disk and regional observation collections

ABI: Designed for calibration



Agenda



ABI-Class Imager Description On-board Radiometric Calibration Targets Operational Calibration Additional Capabilities for Calibration & Validation

Summary



AHI-8 True Color (RGB) Image 3Dec2015 12:00 JST (Data from JMA, Processing by Harris)

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ABI-Class Imager is a Mature Product Line



Supporting three missions:

• GOES-R (ABI), Himawari (AHI), GEO-KOMPSAT-2A (AMI)

Four flight models delivered

- ABI PFM: Integrated on GOES-R spacecraft
- AHI-8: Operating on orbit (Himawari-8)
- ABI FM2: Delivered
- AHI-9: Integrated on Himawari-9 spacecraft

Three more in production at Harris

• ABI FM3, ABI FM4, AMI











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ABI's 2-Mirror Scanner Key to Improved Calibration Capability



- Scans parallel to equator without rotating image
 - 100% scan coverage efficiency
- Lowest inertia and power
- 2x EW <u>and</u> NS mechanical-tooptical motion
- Inherently polarization compensating
 - At nadir, polarization introduced by reflection off NS scanner is canceled by reflection off of EW scanner
 - Blackbody located anti-nadir, so same observing geometry applies



Delivers fast slews and accurate slow scans with minimal disturbance

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ABI Optical Architecture: Simple Solution to Mission Needs





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ABI-class Imagers Provide 16 Channels, Optimized for Customer's Mission

		Resolution	Center wavelength (µn		
FPM	FPA	(km)	ABI	AHI	AMI
VNIR	A047	1	0.47	0.47	0.47
	A064	0.5	0.64	0.64	0.64
	A086	1	0.865	0.51	0.51
	A138	2	1.378	1.61	1.378
	A161	1	1.61	0.865	0.865
	A225	2	2.25	2.25	1.61
	A390	2	3.9	3.9	3.9
	A618	2	6.185	6.185	6.185
MWIR	A695	2	6.95	6.95	6.95
	A734	2	7.34	7.34	7.34
	A850	2	8.5	8.5	8.5
	A961	2	9.61	9.61	9.61
LWIR	A1035	2	10.35	10.35	10.35
	A1120	2	11.2	11.2	11.2
	A1230	2	12.3	12.3	12.3
	A1330	2	13.3	13.3	13.3

- AHI & AMI added 1-km 0.51 µm channel (green)
 - Silicon detector array
 - Improved ocean images
- Retained 1-km 0.865 µm channel
 - Shifted to HgCdTe detector array
- Changed 1.61 µm channel to 2-km
- Eliminated one NIR channel
 - AHI: 1.378 µm
 - AMI: 2.25 µm

Color Key:	
Not in ABI	
Different FPA	

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Detector Selection Capability Provides Operational Redundancy

Requirement: one operational element per downlinked row per side

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Channels	Resolution	EW ASD	IFOV	(µrad)		
(wavelengths in µm)	(km)	(µrad)	NS	EW	Rows	Columns
0.64	0.5	11	10.5	12.4	1460	3
0.47, 0.86	1	22	22.9	22.9	676	3
1.61	1	22	22.9	22.9	676	6
1.38, 2.25	2	44	42	51.5	372	6
3.9, 6.18, 6.95, 7.34, 8.5, 9.61	2	44	47.7	51.5	332	6
10.35, 11.2, 12.3, 13.3	2	44	38.1	34.3	408	6

ASD = angular sample distance

• EW interval at which samples are collected at standard scan rate

NS ASD = NS IFOV (100% fill factor)

• NS interval at which samples are collected

Resolution = pixel spacing of final image after resampling

• 1 km = 28 µrad

77,400 detector elements total; 7,856 downlinked

Large number of detector elements

- Much more to be calibrated
- Increased risk of striping

Large FOV

• Much larger than traditional vicarious calibration scenes

Greater calibration accuracy expectations

Parameter	Units	GOES-O Imager	GOES-R ABI	Ratio
Channels		5	16	3.2
Detector Elements: total		24	77,400	3225
Detector Elements: downlinked		16	7,856	491
NS FOV: max channel	µrad	274	16,311	60
EW FOV: max FPM	µrad	140	33,203	237

Harris' ABI-class imager provides calibration solutions

ABI-Class Imager Description

On-board Radiometric Calibration Targets

Operational Calibration Additional Capabilities for Calibration & Validation

Summary

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Internal Calibration Target (ICT) Accurately Calibrates Emissive Channels On-orbit

- 3-bounce blackbody based on patented 5-bounce Harris design
 - Trap configuration and specular black paint guarantees very high emissivity (>0.995)
 - Robust against stray light and contamination
 - -NIST-traceable
- Built, tested, and demonstrated

ABI PFM 3-bounce blackbody

Full aperture, end-to-end calibration

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Solar Calibration Assembly Delivers On-Orbit Calibration Over Mission Life

- Optical Port Cover:
 - One time deployable
 - Keeps payload clean during launch and outgassing
- Solar Cal Cover (SCC):
 - Open only when calibrating
 - Closed rest of time to preserve cleanliness
- Solar Calibration Target (SCT) is Spectralon[™] diffuser
 - Calibration can occur any day of year at 6:00 a.m. (6:00 p.m. if yaw flipped)
 - Collected with 10x integration time to obtain ~100% albedo signal with sub-aperture target

Optical Port Cover (closed)

SCC (closed)

End-to-end calibration

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Solar Calibration Subsystem Built, Tested, and Qualified for ABI

Design optimized for minimum calibration uncertainty

Optical Port Sunshield Assembly Stray Light Baffles

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End-to-end test collects radiance from on-board targets (ICT & SCT) while varying integration time multiplicative factor

- 0.5x to 16.5x in 33 steps of 0.5x
- 0.0625x to 2x in 32 steps of 0.0625x (1/16th)
- 1x to 22x in 22 steps of 1x

Integration time proportional to integrated photons

More easily controlled than injected voltage levels and tests much more of analog-to-digital signal processing chain

All integration times collected with all targets and all channels

- First set typically used for λ < 3 μm when viewing SCT
 - Nominal SCT observation performed with integration factor of 10x
- Second set typically used for $\lambda > 5 \ \mu m$ when viewing ICT
- Third set typically used for λ = 3.9 µm when viewing ICT

ECAL can also be used when observing space or any other external scene

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Spacelook collected at least every 30 s

- First data collected in every operational timeline
- Automatically collected as part of every Full Disk swath
 - Either at start or end, depending upon scan direction and location of sun
 - Can be autonomously collected on side opposite the sun

Blackbody (ICT) observed at start of each timeline

- Hence, collected at least every 15 minutes
- Ensures all imagery collected during timeline can be radiometrically calibrated

Solar calibration scheduled when needed

- Frequency a function of throughput loss
 - Twice a week at start of mission, once a quarter by end of mission
- ABI: Interrupts operational image collection (uses macro)
- AHI: Special timeline without Full Disk (collects Japan and RO3)

Spacelook subtracted to obtain relative counts Linear coefficient (M) obtained from on-board target observation

Quadratic coefficient (Q) based on pre-launch testing

Quite small and could probably be ignored because detectors are inherently linear

$$\left\langle L_{sample} \right\rangle = \frac{M \cdot \Delta x_{sample} + Q \cdot \Delta x_{sample}^2 - \Delta L_{EW @ sample}^{eff} - \Delta L_{NS @ sample}^{eff}}{\rho_{EW @ sample} \cdot \rho_{NS @ sample}}$$

 $\Delta x_{sample} = \text{scene sample minus spacelook}$ $\Delta L^{eff}_{EW @ sample}, \Delta L^{eff}_{NS @ sample} = \text{contribution from scan mirrors}$ $\rho_{EW @ sample}, \rho_{NS @ sample} = \text{reflectance of scan mirrors}$

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GOES N Imager Boustrophedonic ("as the ox plows")

<u>GOES-R ABI</u> Default = Raster Scan Capable of boustrophedonic

Raster scan results in higher quality images

Constant time interval across swath boundary

Only possible because of Harris' advanced scanner

• Smooth, fast slews at low power with little spacecraft disturbance

ABI baseline scan rate = 0.024434 rad/s (1.4 degree/s)

Scan rate can be increased or decreased if different EW ASD is desired

Harris' ABI Offers Unique Scan Flexibility

NS swath & stare support vicarious calibration for GSICS Angled swaths can compensate for spacecraft yaw

- Utilized by OMC (orbit motion compensation)
- FPM is not rotated; hence coverage decreases as tilt increases

All scenes and timelines can be updated in orbit

Standard calibration scenes easier to observe with ABI Unique collections possible

• Significantly increases types of collections for independent assessments of calibration accuracy

Calibration interleaved with standard scenes

- Greatly increases opportunities for calibration
- Calibration collections should be routine rather than special events

Six examples will be discussed:

- 1. North-South (NS) scan
- 2. Moon scan
- 3. Fine scale raster scan
- 4. Perimeter scans or stares of space
- 5. Simultaneous collections with under-flying payloads
- 6. Simultaneous collections with other ABI-class imagers

Purely NS scan permits every detector element of a single channel to observe same scene

Stable uniform scene provides accurate relative calibration within each channel

• Absolute calibration if source radiance well known

Requires 16 swaths (one per channel)

Swaths tailored to each instrument

- Start and stop match south and north detector line-of-sight
- Swath tilt compensates for FPA rotation
- Duration based on NS IFOV so samples collected at same NS location

Vicarious calibration scene much smaller than NS FOV

 Size based on detector element NS IFOV & FPA EW IFOV, plus pointing uncertainties

Spacecraft yaw compensated by OMC

Collections of different scenes and locations permit effects of scan angles and spectral content to be assessed

North-South Scan Collection Time: 18.99 s

	Wavelength	IFOV	(µrad)			Frame	Duration
FPM	(µm)	NS	EW	Columns	Rows	Time (ms)	(s)
VNIR	0.47	22.9	22.9	3	676	0.902	0.610
	0.64	10.5	12.4	3	1460	0.451	0.658
	0.865	22.9	22.9	3	676	0.902	0.610
	1.378	42.0	51.5	6	372	1.803	0.671
	1.61	22.9	22.9	6	676	0.902	0.610
	2.25	42.0	51.5	6	372	1.803	0.671
MWIR	3.9	47.7	51.5	6	332	1.803	0.599
	6.185	47.7	51.5	6	332	1.803	0.599
	6.95	47.7	51.5	6	332	1.803	0.599
	7.34	47.7	51.5	6	332	1.803	0.599
	8.5	47.7	51.5	6	332	1.803	0.599
LWIR	9.61	47.7	51.5	6	332	1.803	0.599
	10.35	38.1	34.3	6	408	1.803	0.736
	11.2	38.1	34.3	6	408	1.803	0.736
	12.3	38.1	34.3	6	408	1.803	0.736
	13.3	38.1	34.3	6	408	1.803	0.736

Minimum scene (NS x EW): 312 x 577 µrad = 11.1 x 20.6 km

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Moon observations used by many instruments for absolute calibration ABI provides very detailed moon image

- 332 to 1460 rows per FPA
- Moon fills ~60% of ABI NS FOV

Total moon radiance can provide absolute scale for relative calibration obtained from NS scans

- Use relative response to compute weighted total counts for moon image
- Divide into total radiance

Can collect moon as part of timeline

- Interleaved with Full Disk and CONUS
- · Meso time allocation permits two Moon swaths

AHI-8 image courtesy of JMA

Calibration accuracy can be affected by scan angle and variation of angle of incidence over the length of the FPA

- ABI's calibration algorithm includes adjustment for scan mirror angle
- ABI calibrates angle of incidence effects by end-to-end calibration of entire FPA in single observation of on-board calibration target

Effectiveness of on-board target calibration can be validated via a fine raster scan of a stable scene

- Scanning same scene with slight changes in NS swath location
 - E.g., each swath could step south by 280 µrad (10 km), providing 50 swaths for 500 km scene
 - 3 seconds per swath; can be interleaved with Full Disk and CONUS collections

Provides 3rd assessment method of FOV and scan angle effects

 Compare to result of on-board targets and NS scan results

Emissivity of scan mirrors varies with angle of incidence Causes variation in self emission background radiance between spacelook and Earth scene

Component-level measurement of emissivity vs. angle of incidence used as part of on-orbit calibration of scene data

Observing space (zero radiance) at different scan mirror angles provides independent assessment over life of mission

Baseline is to collect data at 8 locations around perimeter of Earth

· Could also do piecewise linear scans

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Comparing images of same scene taken by multiple spacebased payloads provides cross-calibration

ABI provides much greater ability to implement this technique

- More opportunities
 - Collection interleaved with baseline Full Disk and CONUS collections rather than interrupting them
 - Meso opportunity every 30 s in Scan Mode 3 timeline, permitting multiple collections within a single underpass
- Better temporal correlation
 - Meso can be positioned to directly overlap under-flying payload, resulting in simultaneous collections
- Better spectral correlation
 - Sixteen bands to chose between rather than just five

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ABI-class imagers are designed for improved calibration Scan flexibility provides significantly more calibration opportunities and unique methods, including:

- 1. North-South (NS) scan
- 2. Moon scan
- 3. Fine scale raster scan
- 4. Perimeter scans or stares of space
- 5. Simultaneous collections with under-flying payloads
- 6. Simultaneous collections with other ABI-class imagers

Dr. Paul C. Griffith Harris Space and Intelligence Systems paul.griffith@harris.com Appreciation to Tim Schmit, NOAA/NESDIS/STAR/CRPD/ASPB, and to NOAA, JMA, and UW/SSEC CIMSS

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